

Operation status for the asteroid explorer, Hayabusa2

September 27, 2018

JAXA Hayabusa2 Project



Topics



Regarding Hayabusa2:

- MINERVA- II 1 newsflash
- MASCOT release operation
- Images of Ryugu



Contents



- 0. Hayabusa2 & mission flow outline
- 1. Project status & overall schedule
- 2. MINERVA- II 1 newsflash
- 3. MASCOT release operation
- 4. Images of Ryugu
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- 6. Future plans



Overview of Hayabusa2



Objective

We will explore and sample the C-type asteroid Ryugu, which is a more primitive type than the S-type asteroid Itokawa that Hayabusa explored, and elucidate interactions between minerals, water, and organic matter in the primitive solar system. By doing so, we will learn about the origin and evolution of Earth, the oceans, and life, and maintain and develop the technologies for deep-space return exploration (as demonstrated with Hayabusa), a field in which Japan leads the world.

Expected results and effects

- •By exploring a C-type asteroid, which is rich in water and organic materials, we will clarify interactions between the building blocks of Earth and the evolution of its oceans and life, thereby developing solar system science.
- Japan will further its worldwide lead in this field by taking on the new challenge of obtaining samples from a crater produced by an impacting device.
- •We will establish stable technologies for return exploration of solarsystem bodies.

Features:

- ·World's first sample return mission to a C-type asteroid.
- ·World's first attempt at a rendezvous with an asteroid and performance of observation before and after projectile impact from an impactor.
- •Comparison with results from Hayabusa will allow deeper understanding of the distribution, origins, and evolution of materials in the solar system.

<u>International positioning:</u>

- Japan is a leader in the field of primitive body exploration, and visiting a type-C asteroid marks a new accomplishment.
- •This mission builds on the originality and successes of the Hayabusa mission. In addition to developing planetary science and solar system exploration technologies in Japan, this mission develops new frontiers in exploration of primitive heavenly bodies.
- •NASA too is conducting an asteroid sample return mission, OSIRIS-REx (launch: 2016; asteroid arrival: 2018; Earth return: 2023). We will exchange samples and otherwise promote scientific exchange, and expect further scientific findings through comparison and investigation of the results from both missions.



(Illustration: Akihiro Ikeshita)

Hayabusa 2 primary specifications

Mass Approx. 609 kg Launch 3 Dec 2014 Mission Asteroid return Arrival 27 June 2018

Earth return 2020

Stay at asteroid Approx. 18 months

Target body Near-Earth asteroid Ryugu

Primary instruments

Sampling mechanism, re-entry capsule, optical cameras, laser range-finder, scientific observation equipment (near-infrared, thermal infrared), impactor, miniature rovers.



Mission Flow



Launch 3 Dec 2014

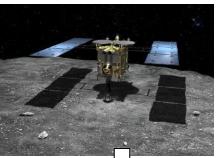


Arrival at asteroid

June 27, 2018

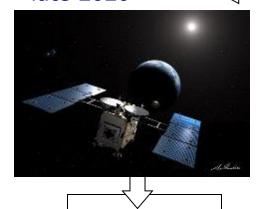
Earth swing-by 3 Dec 2015





Examine the asteroid by remote sensing observations. Next, release a small lander and rover and also obtain samples from the surface.

Earth return late 2020



Depart asteroid Nov-Dec 2019



After confirming safety, touchdown within the crater and obtain subsurface samples



Use an impactor to create an artificial crater on the asteroid's surface

(Illustrations: Akihiro Ikeshita)

Sample analysis



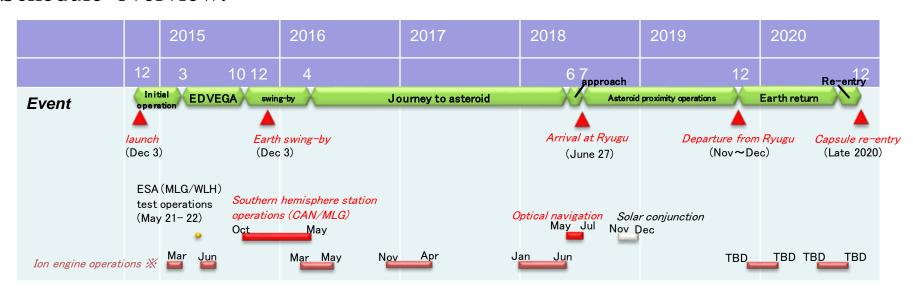
1. Current project status & schedule overview



Current status:

- September 19 21 was the MINERVA-II1 deployment operation, with MINERVA-II1 Rover-1A & Rover-1B separating from the spacecraft on September 21. The two rovers were later confirmed to have landed on the surface of Ryugu and are moving via hopping.
- September 30 October 4 is the MASCOT deployment operation, including the preparatory period.

Schedule overview:







History from deployment to present

- 9/21: 13:05JST separation from the Hayabusa2 spacecraft
- 9/21: Day 1 (Sol.1) Image data obtained after rover deployment
- 9/21: Day 2 (Sol.1) Rover ON confirmation, telemetry pending
- 9/22: Day 3 (Sol.3) Rover ON confirmation, telemetry pending
- 9/22: Day 4 (Sol.4)
 - Rover-1A confirm hopping
 - Rover-1B telemetry received, surface temperature measurement, hop preparation
- 9/23: Day 7 (Sol.7) Rover-1A, 1B telemetry received
 - Rover-1B confirm hopping
- 9/27: Day 20 (Sol.20)





Image taken immediately after separation



Image captured by Rover-1A on September 21 at around 13:08 JST. This is a color image taken immediately after separation from the spacecraft. Hayabusa2 is at the top and the surface of Ryugu is bottom. The image is blurred because the shot was taken while the rover was rotating. (Image credit: JAXA)





Image taken immediately after separation



Image captured by Rover-1B on September 21 at around 13:07 JST. This color image was taken immediately after separation from the spacecraft. The surface of Ryugu is in the lower right. The top left colored blur in the top left is due to the reflection of sunlight when the image was taken. (Image credit: JAXA) we deleted halation.





Image captured during hopping

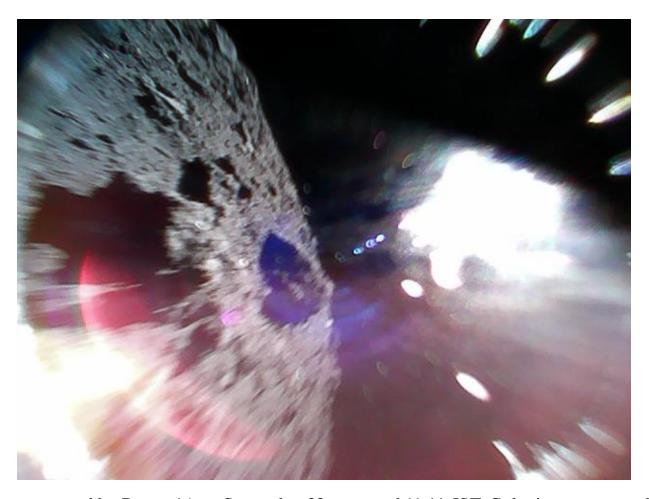


Image captured by Rover-1A on September 22 at around 11:44 JST. Color image captured while moving (during a hop) on the surface of Ryugu. The left-half of the image is the asteroid surface. The bright white region is due to sunlight. (Image credit: JAXA).

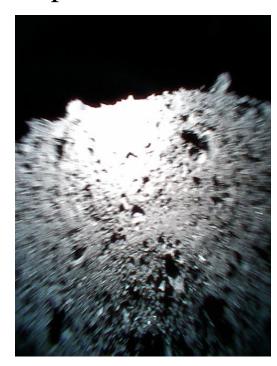




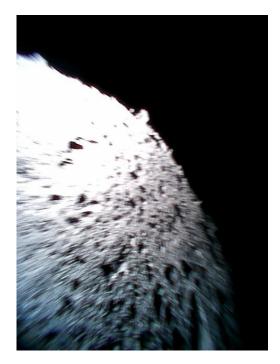
Images taken by Rover-1B

New

September 23, 2018: confirmation of Rover-1B hop



2018/09/23 09:50



2018/09/23 09:55



2018/09/23 10:00

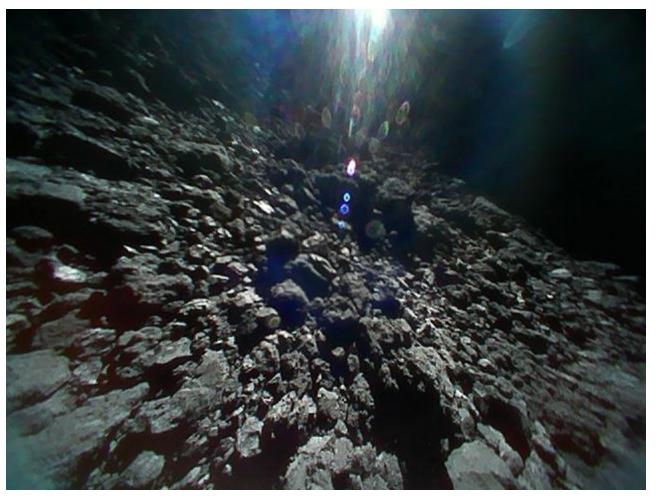
(Image credit: JAXA)





New

September 23, 2018: image captured immediately before hop of Rover-1B

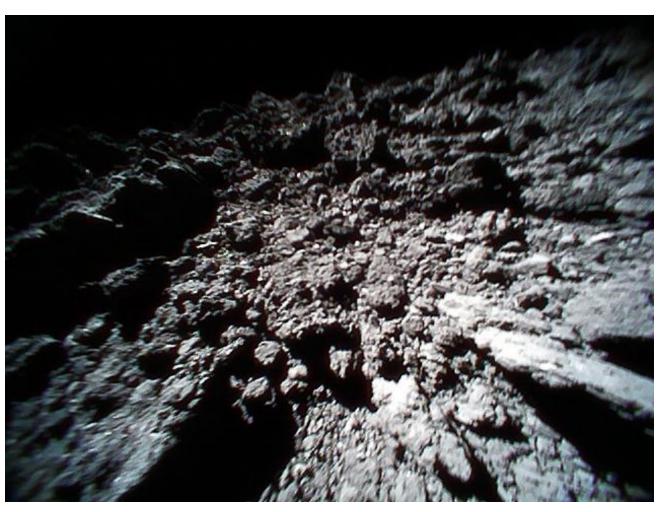






New

September 23, 2018 at 10:10 JST: surface image from Rover-1B after landing







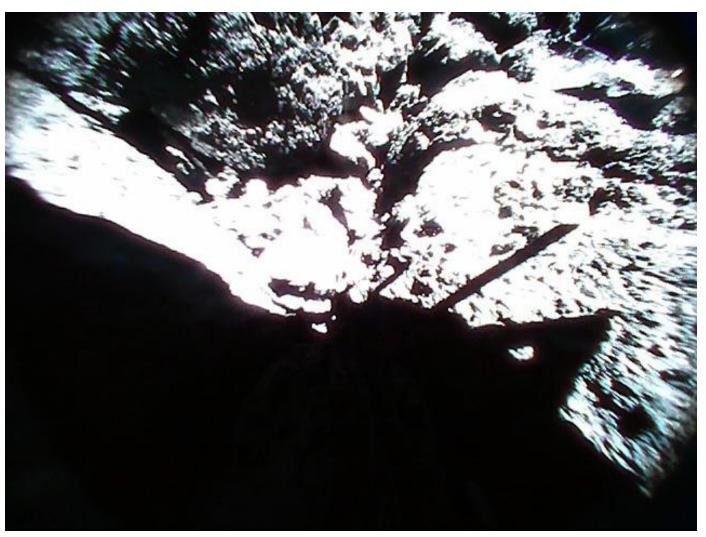
September 23, 2018 at 09:43 JST: surface image taken from Rover-1A



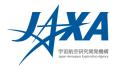
New



September 23, 2018 at 09:48 JST: surface image taken from Rover-1A



New



September 23, 2018 at 09:48 JST: surface image taken from Rover-1A MINERVA-II1 successfully captured the shadow of its own antenna and pin



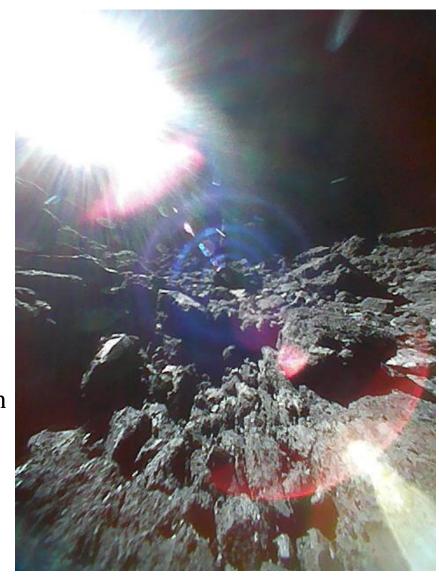
16





Rover-1B successfully shot a movie

New



15 frames captured on September 23, 2018 from 10:34 – 11:48 JST

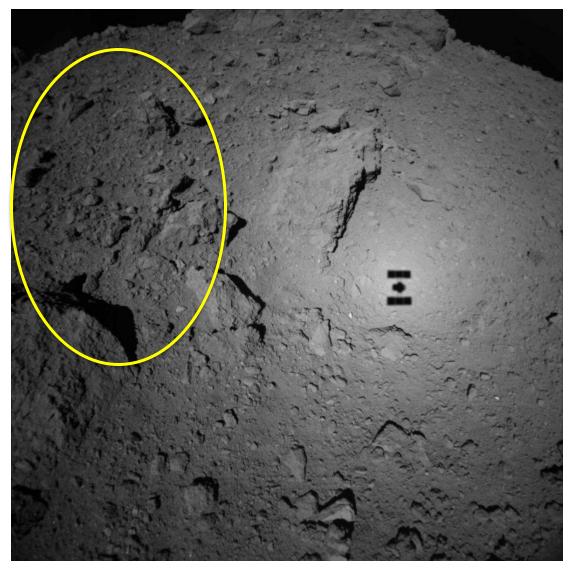
(animation)

(credit:JAXA)



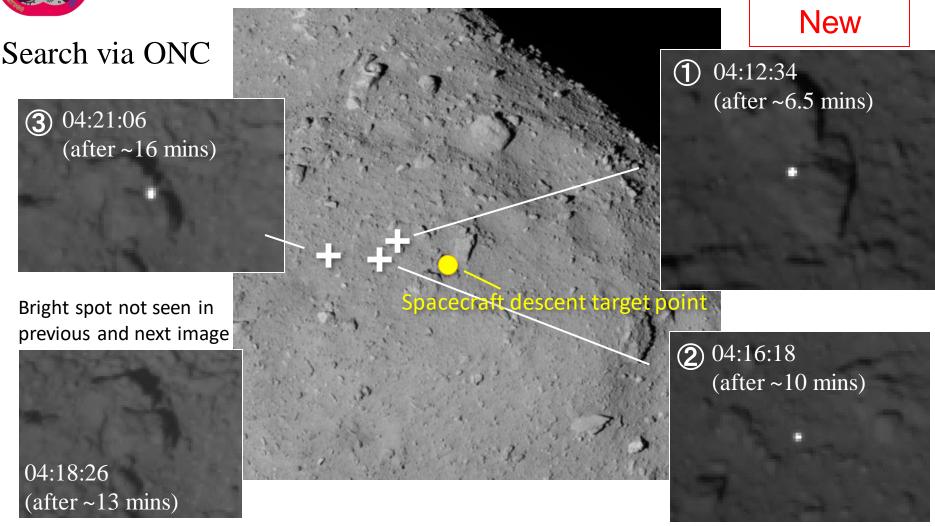
September 21, 2018 at 13:02 JST: image from ONC-W1 at 70m altitude

Separation location of MINERVA-II1 (estimated)









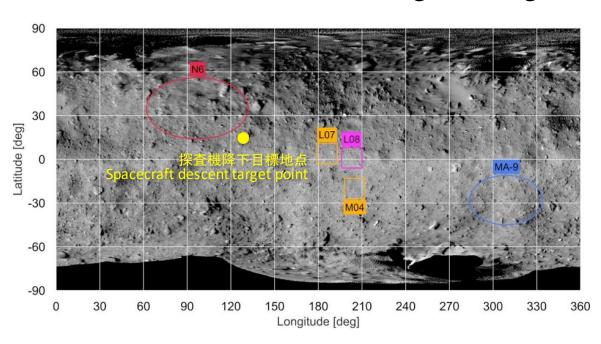
The bright point is confirmed as the MINERVA-II or the container cover moving across $1 \rightarrow 2 \rightarrow 3$, (1, 2): ONC-W1, 3: ONC-T).

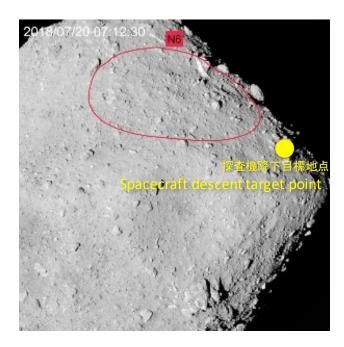
(Image credit: JAXA, AIST, U. Tokyo, Kochi U, Rikkyo U, Nagoya U, Chiba Institute of Tech, Meiji U, U. Aizu)



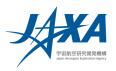


Reference: Location of landing site targets





If MINERVA-II1 separated at the spacecraft descent target point, then post separation MINERVA-II1 should be within the area N6.

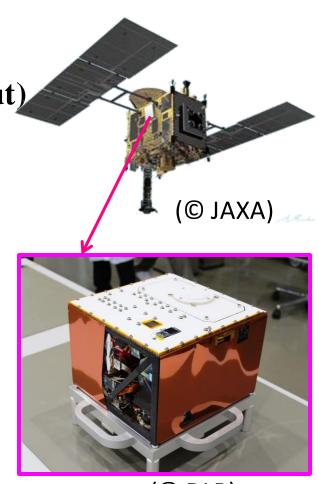




MASCOT System Overview

MASCOT (Mobile Asteroid Surface Scout)

- Developed by DLR (German Aerospace Center) in close cooperation with CNES (French Space Agency)
- Agile, lightweight & compact landing platform for in-situ asteroid research
- Lander Module mass: ~9.8 kg
- Lander Module size: 0.275 x 0.290 x 0.195 m
- Carries Four Scientific Payloads: MASCAM, MicrOmega, MARA, and MASMAG



Flight Model (© DLR)

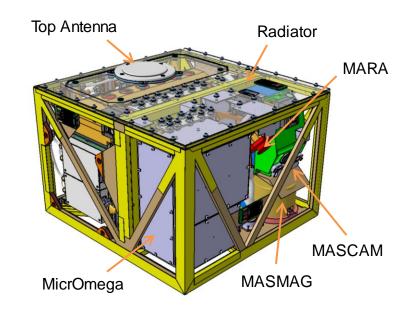




MASCOT System Overview

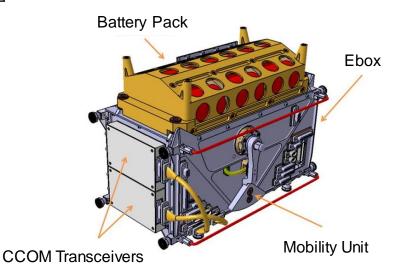
Scientific instruments aboard MASCOT

Device	Function
Wide-angle camera (MASCAM)	Imaging at multiple wavelengths
Spectroscopic microscope (MicrOmega)	Investigation of mineral composition and characteristics
Thermal radiometer (MARA)	Surface temperature measurements
Magnetrometer (MASMAG)	Magnetic field measurement



MASCOT Bus System

- Power: Primary lithium battery
- Communication: Communication system using transceivers same as Minerva-II rovers
- Mobility: Up-righting and hopping mechanism using motor and excenter mass
- GNC: MASCOT attitude determination using proximity sensors

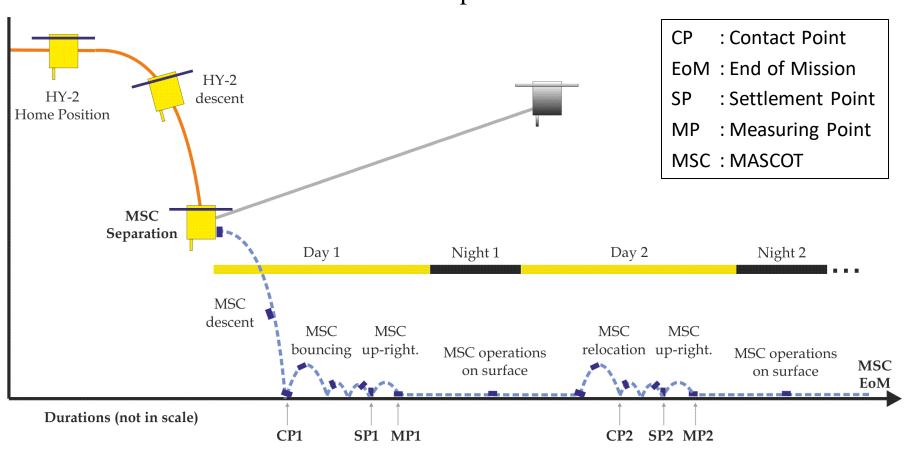






MASCOT On-Asteroid Operation

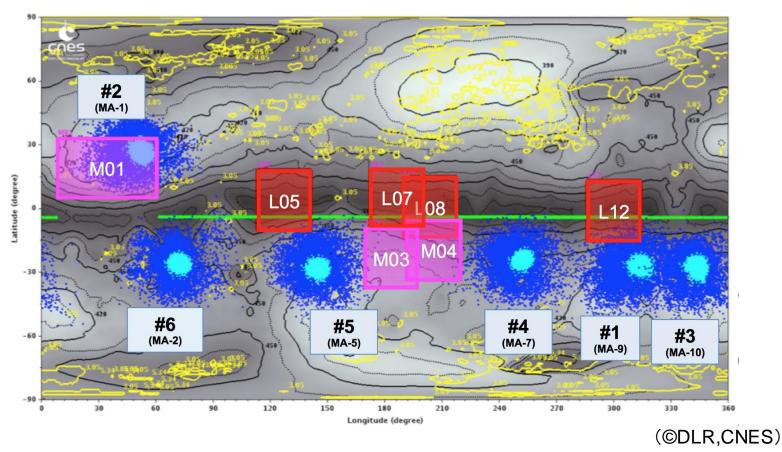
Baseline MASCOT Activities after the Separation



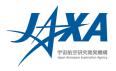




Landing site candidates for MASCOT: Selection from 10 candidates and the order of priority

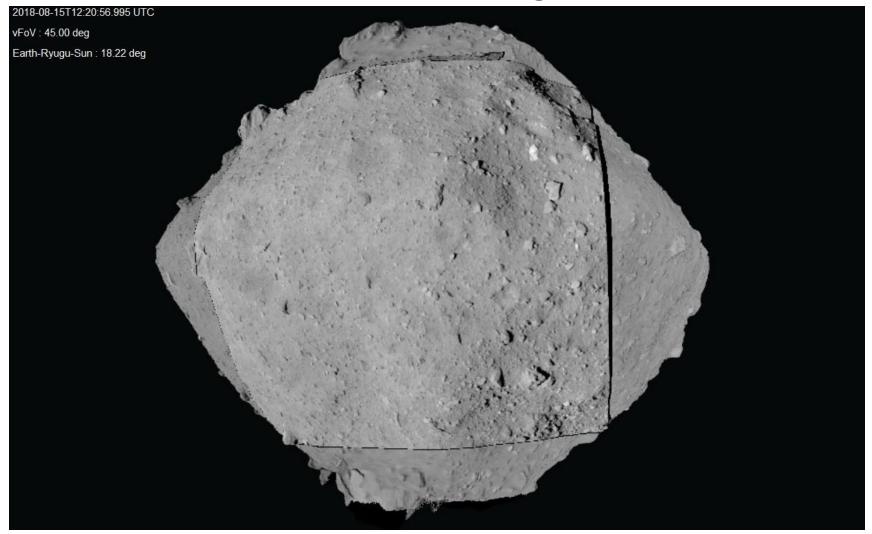


Priority of landing site: MA-9 > MA-1 > MA-10 > MA-7 > MA-5 > MA-2

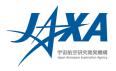




MASCOT landing site

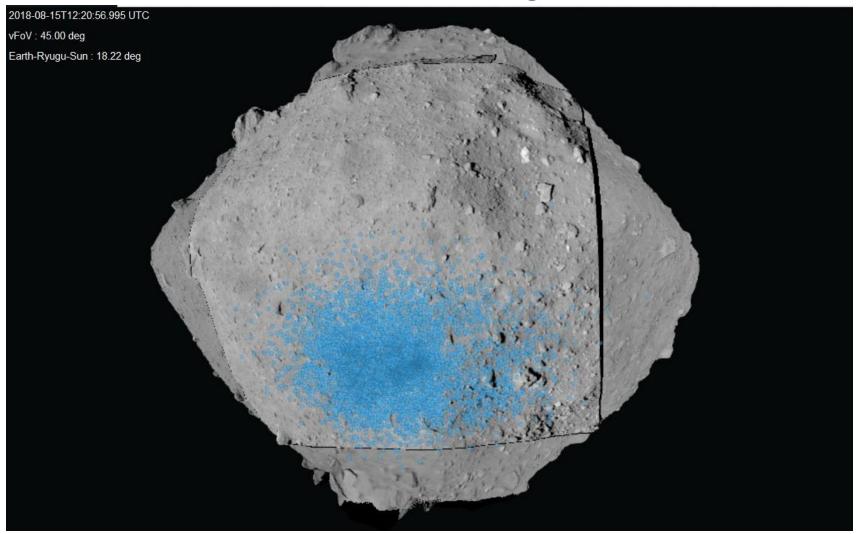


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MASCOT landing site

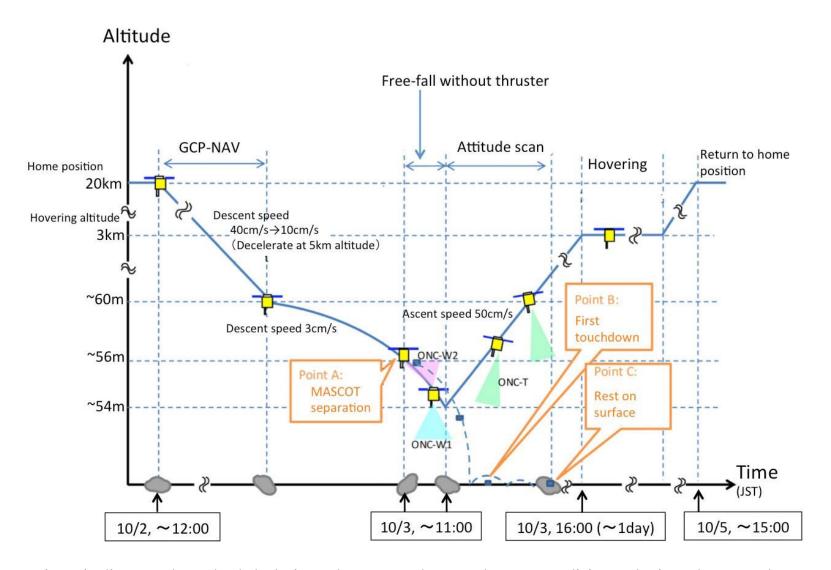


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Outline of MASCOT release operation sequence



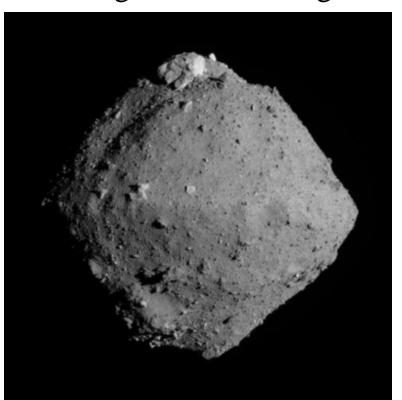
Note: Time indicates the scheduled time, but may change due to conditions during the actual operation.

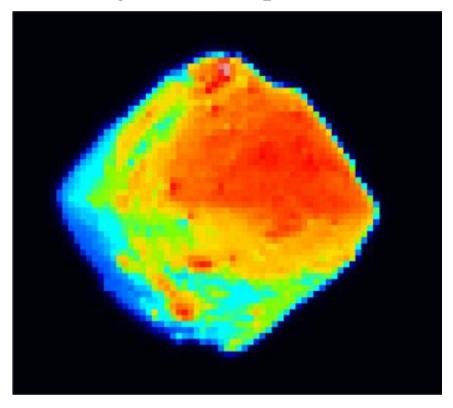


New



Image in the evening direction during BOX-B operations





Images taken in the evening direction of Ryugu. Left is the photograph with the Optical Navigation Camera Telescopic (ONC-T) and on the right is the photograph from the Thermal Infrared Imager (TIR). The photographs were taken on August 31, 2018 at around 19:00 JST from a position of around -x = 9km.

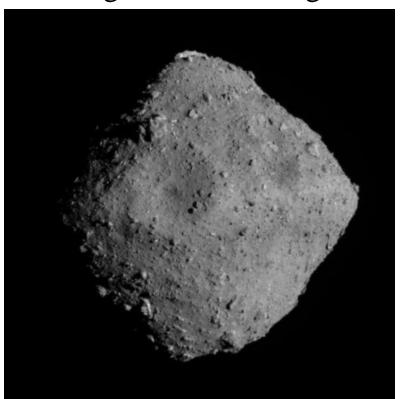
(Left image credit: JAXA, University of Tokyo, Kochi University, Rikkyo University, Nagoya University, Chiba Institute of Technology, Meiji University, University of Aizu, AIST; Right image credit: JAXA, Ashikaga University, Rikkyo University, Chiba Institute of Technology, University of Aizu, Hokkaido University of Education, Hokkaido Kitami Hokuto High School, AIST, National Institute for Environmental Studies, University of Tokyo, German Aerospace Center (DLR), Max Planck Society for the Advancement of Science, Stirling University.)

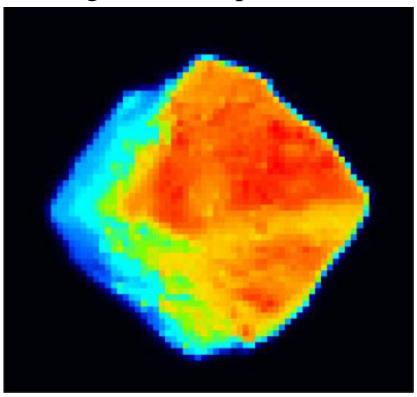


New



Image in the evening direction during BOX-B operations





Images taken in the evening direction of Ryugu. Left is the photograph with the Optical Navigation Camera Telescopic (ONC-T) and on the right is the photograph from the Thermal Infrared Imager (TIR). The photographs were taken on August 31, 2018 at around 23:00 JST from a position of around -x = 9km.

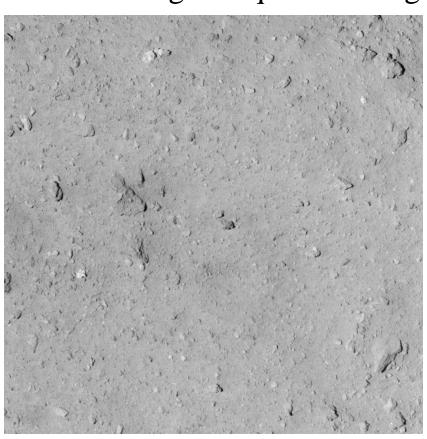
(Left image credit: JAXA, University of Tokyo, Kochi University, Rikkyo University, Nagoya University, Chiba Institute of Technology, Meiji University, University of Aizu, AIST; Right image credit: JAXA, Ashikaga University, Rikkyo University, Chiba Institute of Technology, University of Aizu, Hokkaido University of Education, Hokkaido Kitami Hokuto High School, AIST, National Institute for Environmental Studies, University of Tokyo, German Aerospace Center (DLR), Max Planck Society for the Advancement of Science, Stirling University.)







Images acquired during Touchdown 1 Rehearsal 1



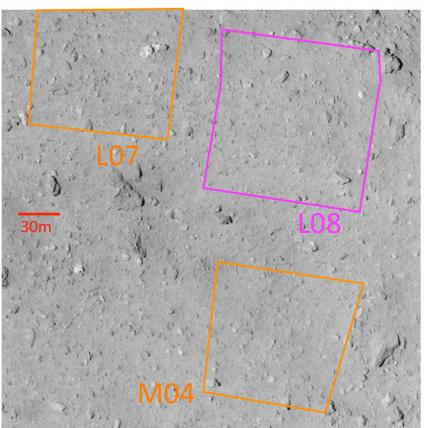


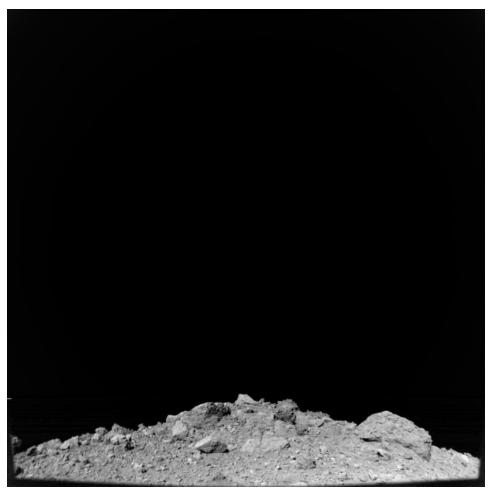
Image of the touchdown candidate sites taken from about 3km above the surface using the Optical Navigation Camera – Telescopic (ONC-T). The image was captured on September 12, 2018 at around 06:00 JST. In the right-hand image, the approximate areas of the touchdown candidate sites are marked out.

(Image credit: JAXA, University of Tokyo, Kochi University, Rikkyo University, Nagoya University, Chiba Institute of Technology, Meiji University, Aizu University, AIST).





Image captured by ONC-W1 during the MINERVA-II1 separation operation



New

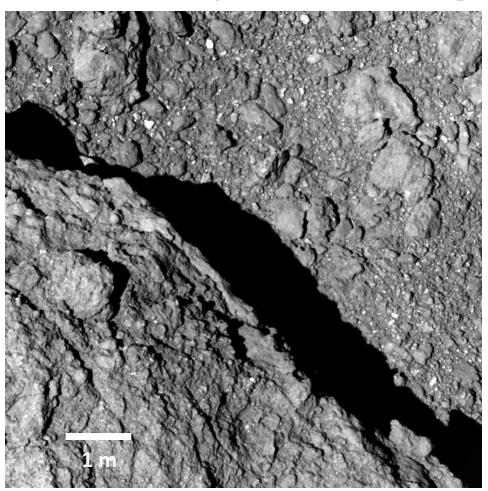
Note: This is the first image taken by ONC-W2.

Image of Ryugu captured with the ONC-W2 at an altitude of about 67m. Image was taken on September 21, 2018 at around 13:00 JST. This photograph of the horizon of Ryugu was taken diagonally downwards from where the W2 is installed on the spacecraft.





Image captured with ONC-T during the MINERVA-II1 separation operation



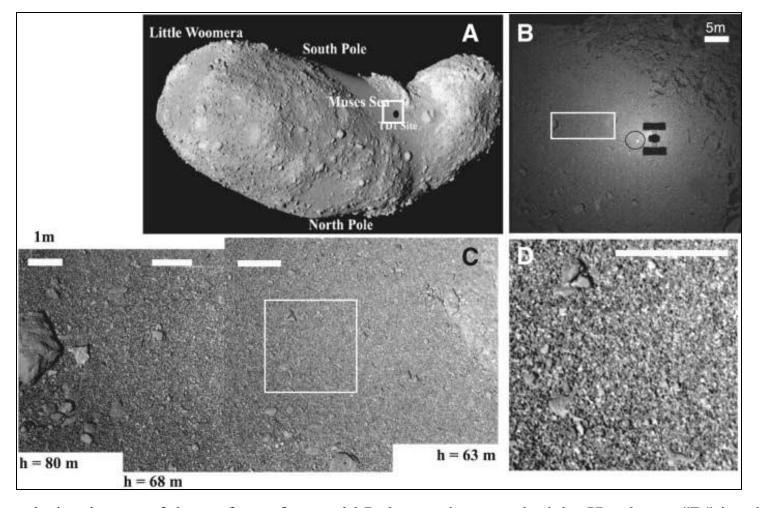
New

Image of Ryugu captured by the ONC-T at an altitude of about 64m. Image was taken on September 21, 2018 at around 13:04 JST. This is the highest resolution photograph obtained of the surface of Ryugu. Bottom left is a large boulder.



Note: Highest resolution photograph obtained by Hayabusa





High resolution image of the surface of asteroid Itokawa photographed by Hayabusa. "D" is taken from an altitude of 63m. It is thought that the so-called "Muses Sea" (official name "MUSES-C Regio") is covered with a "gravel" of granules with diameters from a few mm to few cm. (From Yano et al, Science Vol 312 2, June 2006)



5. Other issues



- On the "Imagining Ryugu" contest
 - Hayabusa 2 Project nodes (organizations and institutes that gathered entries) must now each nominate works.
 - •Final judgement will be at the end of the year.



6. Future Plans



- Operation schedule
 - September 30 October 4 : MASCOT release operation (including preparatory period)

MASCOT separation scheduled for October 3 at around 11:00 JST.

- Mid-October: Touchdown 1 Rehearsal 2
- Late October: Touchdown 1
- Press briefings and media events
 - October 3 (Wednesday) 9:30~18:30 Open press center@Sagamihara 15:00~16:00 Press briefing
 - 17:00~18:00 MASCOT press conference transmission
 - October 11 (Thursday) 15:30~16:30 Press conf.@Ochanomizu

Note: Time is in JST.



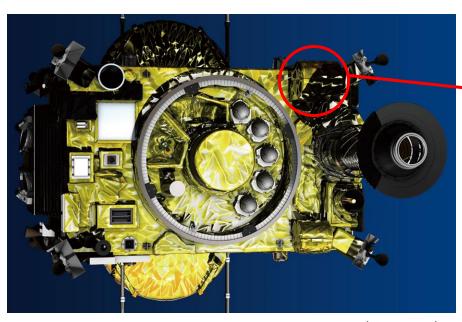


Reference

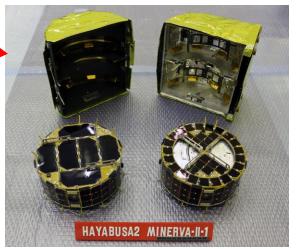




MINERVA-II is the successor of MINERVA installed on the Hayabusa spacecraft.



MINERVA-II1 (Rover-1A, Rover-1B)



Produced at JAXA

(©JAXA)

Collaborating manufacturers, universities and related organizations>

Aichi University of Technology, University of Aizu, Addnics corp., Antenna Giken Co., Ltd., ELNA, CesiaTechno, The University of Tokyo, Tokyo Denki University, Digital Spice Corp., Nittoh Inc., Maxon Japan, DLR, ZARM

- Weight (including deployment)
- MIMERVA-II1:2.5kg
- MINERVA-II1

Twin rovers





MIcro Nano Experimental Robot Vehicle for Asteroid, the Second Generation

- New hopping mobility
- Adaptation with AI
- Small, light-weight,
- Low power consumption
- Autonomous behavior



(©JAXA)

• Scientific observation (stereo sensing, thermometer)





Specifications of Rover-1A, -1B

body size	Cylinder (hexadecagonal pole) Diameter: \phi180[mm] Height: 70[mm]
weight	1A:1151[g], 1B:1129[g]
actuators	DC motor
sensors	4 cameras(1A), 3 cameras(1B) photodiodes, accelerometer thermometers, gyro
com.	32k[bps](max)



Selection of landing site candidates for MINERVA- II



Conditions for MINERVA-II landing site selection:

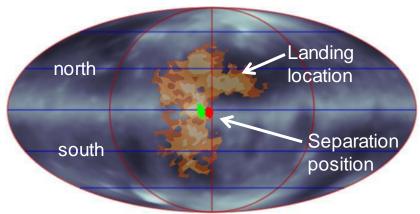
- Landing site does not overlap with spacecraft touchdown candidates.
- Landing site does not overlap with MASCOT landing site candidates.
- ☐ The altitude of the spacecraft after separation must not be lower than 30m.
- Ensure communication with ground station.
- Ensure communication with Hayabusa2 spacecraft.
- Not high temperature region, and fewer parts in shadow

• Due to the equatorial ridge, separation near the equator results in widely spaced landing points to the north and south.

• Separating in the southern hemisphere may result in a spacecraft altitude below 30m.



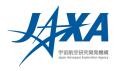
•Separate in northern hemisphere, more than 100m north of the equator.



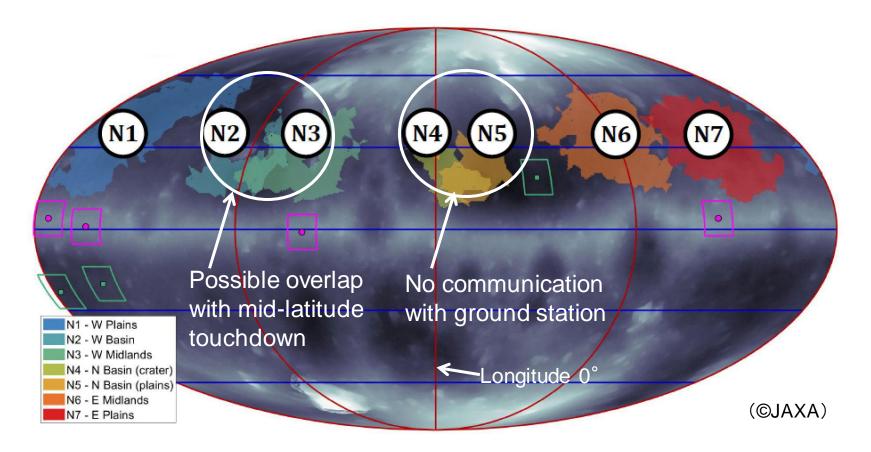
(©JAXA) When separating near the equator, the landing position spread north and south.



Selection of landing site candidates for MINERVA- II



Landing site candidates for MINERVA-II: northern hemisphere



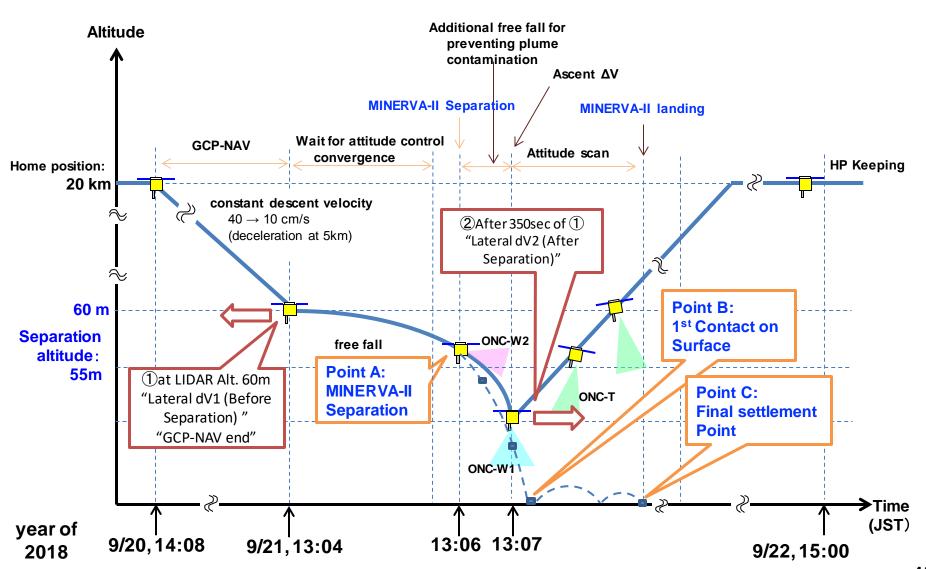
- •Touchdown confirm no overlap with MASCOT's landing site.
- Also consider observability etc. using the ONC-T camera.







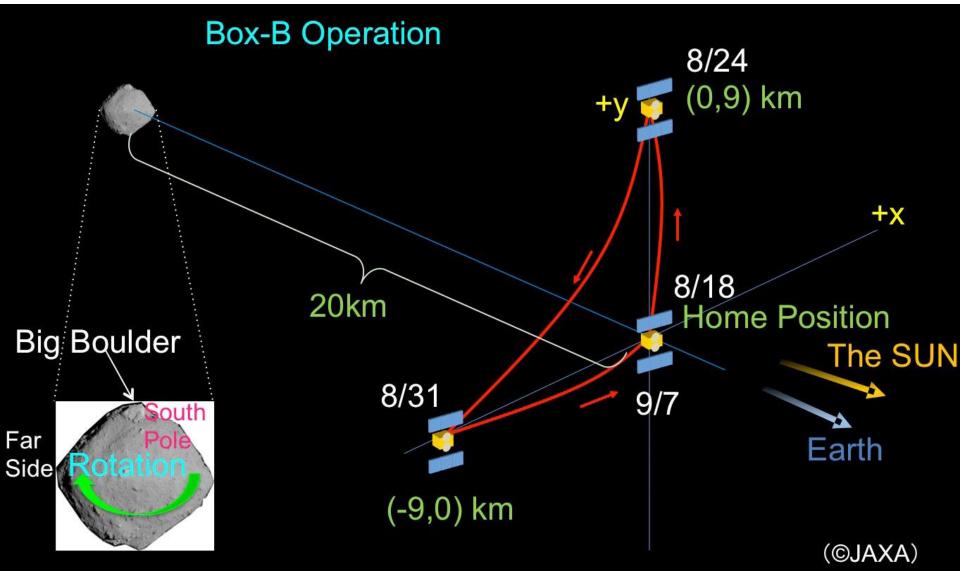
MINERVA- II 1 schematic showing the deployment operation sequence





BOX-B Operation



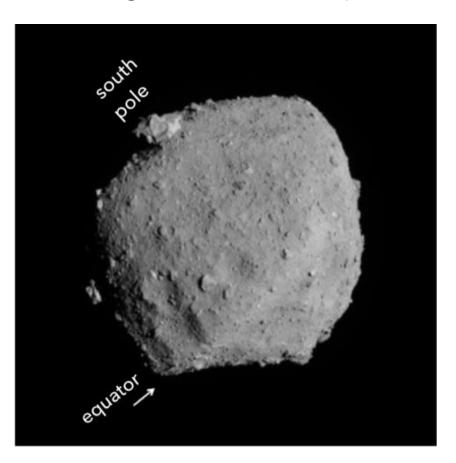




BOX-B operation images



Image of the south pole



Published

(but south pole is now at the image top)

Image of Ryugu towards the asteroid's south pole. Photograph taken with the Optical Navigation Camera Telescopic (ONC-T) on August 24, 2018 at around 17:00 JST. The image was captured from a position of around +y = 9 km. The distance to Ryugu was about 22 km.

(Image credit: JAXA, University of Tokyo, Kochi University, Rikkyo University, Nagoya University, Chiba Institute of Technology, Meiji University, University of Aizu, AIST)



Touchdown 1 Rehearsal 1



TD1-R1 operation overview (Time is JST)

September 11, 2018

15:27 Start descent from home position (altitude 20km), descent speed 0.4m/s

20:20 Altitude approximately 10km

September 12

02:05 Descent slowed to 0.1m/s

10:40 Altitude approximately 1.5km

12:57 Descent cancelled at an altitude of about 600m^{*}, began rising at 0.5m/s

16:30 Acceleration for return to home position

September 13

15:00 Return to home position

*An automatic change was planned from the LIDAR long-distance mode to the short-distance mode. However, it was impossible to reach the required light intensity needed for the switch as the reflected light for distance measurement was too weak. The altitude measurement therefore became impossible and the spacecraft autonomously ceased descent and began to rise.